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Article

# **Emulating Rotational Inertia of Synchronous** Machines by a New Control Technique in **Grid-Interactive Converters**

MDF

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Abstract: Integration of renewable energy sources (RESs) into power systems is growing due to eco-friendly concerns and ever-increasing electricity demand. Voltage source converters (VSCs) are the main interface between RESs and power grids, which have neither rotational inertia nor damping characteristics. Lack of these metrics make the power grid sensitive to frequency disturbances and thereby under frequency, to load shedding activation or even large-scale collapse. In this regard, the contribution of this paper is to develop a new control technique for VSCs that can provide virtual inertia and damping properties with the DC-link capacitors inhered in the DC-side of grid-tied VSCs. The applied VSC is controlled in the current controlled model, with the capability of injecting extra active power with the aim of frequency support during perturbations. The dynamics assessment of the proposed platform is derived in detail. It is revealed that the control scheme performs in a stable region even under weak-grid conditions. Finally, simulations are conducted in MATLAB to depict the efficacy and feasibility of the proposed method. The results show that frequency deviation is mitigated under step up/down changes in the demand, and the rate of change of frequency is improved by 47.37% compared to the case in which the synthetic inertia loop is canceled out.

Keywords: synthetic inertia; frequency stability; virtual synchronous machine; grid-connected voltage source converter

### 1. Introduction

In conventional power grids, the major portion of electricity demand is generated by synchronous generators (SGs), which have inherent rotational inertia and damping properties. During major disturbances (typically loss of generating unit or stochastic demand fluctuation), primary and secondary controls are triggered to maintain stable system frequency and restore power balance [1]. The kinetic energy preserved in the rotational mass of the SGs is highly important in the primary control, which imposes a limit on the frequency oscillations and slows down frequency dynamics [2,3]. Lack of this energy in low-inertia grids might lead to undesirable situations such as pole slipping and catastrophic failure of SGs under frequency load shedding activation or even cascading failure after perturbation. The power system can be more vulnerable to these problems under high levels of renewable energy sources (RESs) penetration [4,5]. In other words, the power electronics-based generators have no rotational inertia and damping characteristics. The frequency regulation adequacy is assessed by two more cited metrics: (1) frequency nadir or the minimum frequency reached following a disturbance and (2) rate of change of frequency [3]. The overall grid inertia is the main factor affecting

the aforesaid indices, i.e., the more inertia, the better primary frequency regulation [6]. On the other hand, the diminution of grid inertia limits further increase in RESs integration. To address this issue, grid-connected converters, such as the interface between RESs and power systems, must be equipped with grid supportive functionalities defined by new grid codes [7,8].

Intensive research has been conducted on the control of grid-tied converters with respect to the aforementioned concerns [9–22]. Changing the requirements of rate of change of frequency (RoCoF) withstand capabilities of SGs is a straightforward solution that has been accepted as an efficient approach by system operators in Ireland [9]. However, the inertia concern is not solved in this method and the cost increment with regard to generator testing might bind its applications. Employing backup SGs with partial loading (i.e., spinning reserve) or synchronous condensers to provide more rotating masses is another possible solution for inertia augmentation [10]. Nevertheless, these approaches yield high capital and operating cost. Motivated by conventional SG dynamics, the concept of providing synthetic inertia using virtual synchronous generators (VSGs) VSGs was introduced in 2007 [11]. This method emulates the inertial response of real SGs by the grid-connected voltage source converters (VSCs) through a control technique, resulting in fast frequency support during contingencies. Since then, similar concepts, named synchronverter and inducverter, have been developed, on which detailed information can be found in [12,13]. Additionally, [14] improved the power quality and grid stability of distributed generation using VISMA. This technique aims to set up static and dynamic characteristics of SGs on the three-phase hysteresis controlled inverters. Similar to conventional generators, wind turbines enjoy a considerable amount of kinetic energy preserved in their blades. However, a variable speed wind turbine generator (e.g., doubly-fed induction generator or permanent magnet synchronous generator) does not naturally contribute to the grid inertia due to the converter interface [15]. Various control mechanisms have been introduced to enable wind turbines to provide inertia support. For example, an enhanced control technique is proposed in [16] that adjusts the turbine torque set point as a function of the frequency deviation  $\Delta f$  and the RoCoF, enabling wind generator to emulate inertia. Employing energy storage systems (ESS) ESSs such as supercapacitors and battery banks on the DC-side of VSCs and proportional linking their active power references with the frequency oscillations is another emerging method of enhancing inertial response in power systems [17,18]. These approaches can provide sufficient inertia and damping properties for fast restoring power balance in the grid. The research work in [19] designed a high-performance battery converter system for providing virtual inertia at the distributed network level in which the grid frequency is measured to derive design criteria for the control system. The frequency derivative-based (df/dt) method is the other technique of inertia emulation by VSCs. Nonetheless, the research conducted in [20] shows that df/dt-based synthetic inertia emulation can induce instability into two modes of the system, one associated with the output filter and the other one with the inner current control loop. In [21], the authors developed a new technique of emulating virtual inertia through the DC-link capacitors adhered to the DC-side of grid-tied converters. The concerns with respect to this approach are instability problems due to applying df/dt based inertia emulation and the limitation of its responsiveness [20,22].

The main contribution of this work to the research field is summarized as follows: we develop a new control technique for the provision of synthetic inertia support through grid-tied voltage source converters employed in the power grids dominated by RESs. The proposed platform comprises two main control loops, i.e., an inner current loop and outer voltage loop equipped with a new synthetic inertia control. The synchronous reference frame transformation introduced by the phase-locked loop (PLL) dynamics is also included in the control system for the required compensation. It is noteworthy that the DC-link capacitance is applied as the energy buffer for grid inertia augmentation. With the proposed control scheme, (1) the system stability margin and (2) the inertial response of the converter are improved compared to the method in [21].

The rest of this paper is organized as follows. Following Section 1, a brief analysis of inertial response in the real SGs is presented in Section 2. Then, a detailed model of the proposed VSC

deriving closed-loop transfer functions of the system and bode analyses. Furthermore, Section 4 includes the comparison of the proposed method with [21] in terms of system stability margin. Section 5 provides simulations to depict the efficacy of the system under step up/down changes in the demand. The results demonstrate that the proposed controller enhances the RoCoF metric by 47.37% and 23.1% compared to the conventional VSC system and [21], respectively. The paper ends with the conclusion presented in Section 6. controller is derived in Section 3. The stable operation of the system is revealed in Section 4 by deriving

2. Inertial Response of Synchronous Machines comparison of the proposed method with [21] in terms of system stability margin. Section 5 provides

It is involutions to a least of the system water the analyzing the system water the second states of the system of the system water the system of the system describing the power imbalance between input mechanical power and electromagnetic power to the conventional VSC system and [21], respectively. The paper ends with the conclusion presented following a contingency. Thereby, the equation of motion is represented by:

2. Inertial Response of Syn 
$$P_m - P_e - D(\omega - \omega_r) = J\omega \frac{d\omega}{dt} \approx J\omega_r \frac{d\omega}{dt}$$
. (1)

..... by the swing equation [1] It is known that the starting

Note that is the power imbalance between input mechanical power and electromagnetic power following a contingency. Thereby, the equation of motion is represented by: can be rewritten as:

$$P_m - P_e - L \ \omega = \omega_{no} - m P_e^{J_{c}} \approx J \omega_r \frac{d\omega}{dt}.$$
(1)

Note that in (1), it is assumed that  $\lim_{r \to \infty} \omega_r = \omega_r + m P_m$  G is one. During steady state operation,

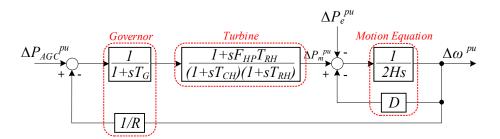
in which  $m^{2} = h/D$  is synonymous with the droop coefficient. It is clear that the primary motion equation-based control is the developed version of simple droop control [23]. After a disturbance, the small-signal model of (1) in terms of inertia constant ( $H = J\omega_r^2/2VA_{base}$ ) is represented by [24]:

equation-based control is the  
the small-signal model of (1) is 
$$\Delta P_m^{pu} - \Delta P_e^{pu} - D\Delta \omega^{pu} = 2H \frac{d\Delta \omega^{pu}}{dt} \cdot \operatorname{rol}[23]$$
. After a disturbance,  
 $tag{ase}$  is represented by [24]: (3)

The frequency control diagram  $A^{p_{u}}$  a  $A^{p_{u}}$  a  $A^{p_{u}}$  iffustrated in Figure 1, and the corresponding parameters are defined in Table 1. Since only the primary frequency regulation and inertial response The frequency control diagram of a real SG is illustrated in Figure P, and the corresponding are of interest in this work the automatic generation control signal (APAG<sup>PU</sup>) is neglected. To analyze the effect of rotational insertia, one the system fraguency response the transferet was to anoty frequency deviation (Attectore the total the isturbances (tar Perisdenive constants), the transfer function of frequency

deviation  $(\Lambda \omega)$  over the load disturbance  $(\Lambda P_{-})$  is derived as (4)

$$TF_{\Delta\omega/\Delta Pe}(s) = \frac{\Delta\omega(s)}{\Delta P_{e}(s)} = \frac{-R(1+sT_{G})(1+sT_{CH})(1+sT_{RH})}{(D+2Hs)(1+sT_{G})(1+sT_{CH})(1+sT_{CH})(1+sT_{RH})R+sF_{HP}T_{RH}+1}.$$
 (4)

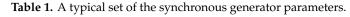


FigureFigureA: frequency response control of any hour segenelator (SG).

The grid frequency response under a 5% step-up change in the demand ( $\Delta P_e = 5\%$ ) is depicted in Figure 2 for different values of H. It is clear that increasing H results in a nover RoCoF level and frequency nadir and, thereby, better primary frequency regulation during the transient time.

1 afameter	value
Governor time constant ( $T_G$ )	0.1 s
Inlet volume time constant ( <i>T</i> CH)	0.2 s
Re-heater time constant ( $T_{RH}$ )	7 s
Turbine coefficient ( <i>F</i> <sub>HP</sub> )	0.3 s
Droop coefficient ( <i>R</i> )	0.05
Inertia constant (H)	5–20 s

Parameter	Value
Sustainability <b>2020</b> , $\frac{1}{2}$	0.1 s
Inlet volume time constant $(T_{CH})$	0.2 s
Reparate in the contribution of the second	71



The grid frequency response  $\overline{\text{Up}Br}_{B}$  a  $\overline{\text{Serificient}}(\overline{Br}_{B})$  is depicted in Figure 2 for different values of H. It is clear that increasing H securits in a lower Rocor level and frequency radii and, thereby, better primary frequency) regulation during the transient time.

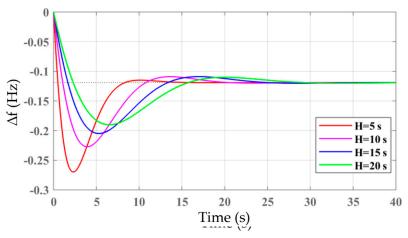


Figure 2. Crid frequency variations for different inertia constants.

# 3. The Proposed Converter Control Scheme

Figure 3 demonstrates the single-line diagram of a grid-tied VSC, comprised of a RES generator, a DE-ink capitoticon contractive to a cuinvalification three phase inhibiting inveitance institution and a valuated power optimal three phase indiverses inveitance interference inter

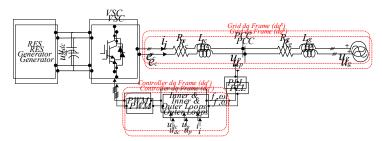


Figure 3. Schematic of the grid-tied voltage source converter (VSO).

3.1. Analysis of the Interfaced Converter, Under, Bynamic Operating Condition 3.4. Analysis of the Interfaced Converter, Under, Bynamic Operating Condition The AC-side of the converter can be described mathematically as:

$$((\underline{\mu}_{p}\pm\gamma))_{i}=\underline{e}_{e_{c}}-\underline{u}_{e_{c}}$$
(55)

where *p* denotes the differential operator,  $L \equiv E_c + E_{s,r}$  and  $r \equiv R_c + R_{s,r}$ . Transforming (3) into the grid synchronous reference frame (*diff*) using Park Transformation (*Twistia*) results in:

$$T_{\text{Edbc/dg}} \equiv \frac{2}{3} \begin{bmatrix} \cos\theta & \cos\left(\theta - \frac{2\pi}{3}\right) & \cos\left(\theta + \frac{2\pi}{3}\right) \\ -\sin\theta & -\sin\left(\theta - \frac{2\pi}{3}\right) & -\sin\left(\theta + \frac{2\pi}{3}\right) \end{bmatrix}$$
(8)

where *p* denotes the differential operator,  $L = L_c + L_g$ , and  $r = R_c + R_g$ . Transforming (5) into the grid synchronous reference frame ( $dq^g$ ) using Park Transformation ( $T_{abc/dq}$ ) results in:

$$T_{abc/dq} = \frac{2}{3} \begin{bmatrix} \cos\theta & \cos\left(\theta - \frac{2\pi}{3}\right) & \cos\left(\theta + \frac{2\pi}{3}\right) \\ -\sin\theta & -\sin\left(\theta - \frac{2\pi}{3}\right) & -\sin\left(\theta + \frac{2\pi}{3}\right) \end{bmatrix}$$
(6)

$$i_{d} = G_{plant}(s) \times \left(e_{cd} - u_{gd} + \omega_{r}Li_{q}\right)$$
  

$$i_{q} = G_{plant}(s) \times \left(e_{cq} - u_{gq} - \omega_{r}Li_{d}\right)$$
(7)

where  $G_{plant}(s)$  is expressed as:

$$G_{plant}(s) = \frac{1}{Ls+r}.$$
(8)

The subscripts *d* and *q* stand for the *d*-axis and *q*-axis components of the corresponding variables. It is clear that the *d* and *q* axes are coupled through the terms  $\omega_r Li_d$  and  $\omega_r Li_{q_r}$  which gives rise to the control complexity. Nonetheless, the cross-coupling terms can be eliminated by applying appropriate current feedforward control. The inner current loops (7) are controlled via two separate proportional integral (PI) PI controllers ( $G_i(s) = k_{cp} + k_{ci}/s$ ) since the *dq* variables are constant in the synchronous reference frame.

Ignoring power losses in the converter, the output current of the VSC can be translated into the DC-link capacitor voltage ( $u_{dc}$ ) by:

$$G_1(s) = \frac{u_{dc}(s)}{i_d(s)} = \frac{-3U_{pd}}{2U_{dc,ref}Cs}.$$
(9)

The voltage  $u_{dc}$  reaches its reference by controlling the *d*-axis current through a PI controller  $(G_u(s) = -k_{up} - k_{ui}/s)$ , which forms the outer control loop. The time-delay introduced by the pulse-width modulation (PWM) PWM updates and reference computations can be approximately modeled using a first-order function  $G_d(s)$  in which  $f_s$  denotes the sampling frequency [25].

$$G_d(s) = \frac{1}{1 + \left(\frac{1.5}{f_s}\right)s}.$$
(10)

### 3.2. Dynamic Analysis of the Phase-Locked Loop

The synchronization of the VSC with the network is obtained using a PLL. Due to the operation of the PLL, the system shown in Figure 3 comprises two separate dq frames:  $dq^g$  frame corresponding to the grid and  $dq^c$  frame corresponding to the VSC controller. In steady state operating conditions, the controller dq frame is aligned with the grid dq frame. When a small contingency occurs in the system, the two frames are no longer aligned due to the dynamics of the PLL. It means that the  $dq^c$  frame lags  $\Delta\theta$  as demonstrated in Figure 4a.

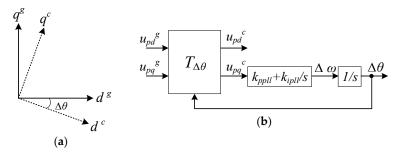


Figure 4. (a) The grid and controller SRFs, (b) SRF-PLL model.

The averaged model of the synchronous reference frame PLL (SRF-PLL) is shown in Figure 4b, in which the PCC voltage vector in the  $dq^g$  frame is rotated to the  $dq^c$  frame using matrix  $T_{\Delta\theta}$  for feedback control purpose [26]. During steady state operation, we can conclude (12):

$$T_{\Delta\theta} = \begin{bmatrix} \cos(\Delta\theta) & \sin(\Delta\theta) \\ -\sin(\Delta\theta) & \cos(\Delta\theta) \end{bmatrix}$$
(11)

$$\underline{u}_p^g = \underline{u}_p^c \tag{12}$$

where the vectors  $\underline{u}_p{}^g$  and  $\underline{u}_p{}^c$  denote the PCC voltage in the grid frame and the controller frame, respectively. Equation (12) points out that the angle difference ( $\Delta\theta$ ) between PCC voltage vectors in the two frames is zero. Using matrix  $T_{\Delta\theta}$ , (12) can be rewritten as:

$$\begin{bmatrix} u_{pd}^{c} \\ u_{pq}^{c} \end{bmatrix} = \begin{bmatrix} \cos(0) & \sin(0) \\ -\sin(0) & \cos(0) \end{bmatrix} \times \begin{bmatrix} u_{pd}^{g} \\ u_{pq}^{g} \end{bmatrix}.$$
 (13)

A disturbance in the grid leads to small signal deviations in the PCC voltage vector. Then, (13) can be rewritten as:

$$\begin{bmatrix} U_{pd}^{c} + \widetilde{u}_{pd}^{c} \\ U_{pq}^{c} + \widetilde{u}_{pq}^{c} \end{bmatrix} = \begin{bmatrix} \cos(0 + \Delta\theta) & \sin(0 + \Delta\theta) \\ -\sin(0 + \Delta\theta) & \cos(0 + \Delta\theta) \end{bmatrix} \times \begin{bmatrix} U_{pd}^{g} + \widetilde{u}_{pd}^{g} \\ U_{pq}^{g} + \widetilde{u}_{pq}^{g} \end{bmatrix}.$$
 (14)

By simplifying trigonometric functions and ignoring the steady state values, the connection between PCC voltage vectors in the  $dq^c$  and  $dq^g$  frames can be expressed as:

$$\begin{bmatrix} U_{pd}^{c} + \widetilde{u}_{pd}^{c} \\ U_{pq}^{c} + \widetilde{u}_{pq}^{c} \end{bmatrix} \approx \begin{bmatrix} 1 & \Delta\theta \\ -\Delta\theta & 1 \end{bmatrix} \times \begin{bmatrix} U_{pd}^{g} + \widetilde{u}_{pd}^{g} \\ U_{pq}^{g} + \widetilde{u}_{pq}^{g} \end{bmatrix} \\ \begin{bmatrix} \widetilde{u}_{pd}^{c} \\ \widetilde{u}_{pq}^{c} \end{bmatrix} \approx \begin{bmatrix} \widetilde{u}_{pd}^{g} + U_{pq}^{g} \Delta\theta \\ -U_{pd}^{g} \Delta\theta + \widetilde{u}_{pq}^{g} \end{bmatrix}.$$
(15)

From Figure 4b,  $\Delta \theta$  is obtained as:

$$\Delta \theta = \widetilde{u}_{pq}^{c} \times \left( k_{ppll} + \frac{k_{ipll}}{s} \right) \times \frac{1}{s}.$$
(16)

Substituting (16) into (15) results in the relationship between  $\Delta \theta$  and  $u^{g}_{pq}$  as:

$$TF_{PLL} = \frac{\Delta\theta}{\tilde{u}_{pq}^g} = \frac{k_{ppll}s + k_{ipll}}{s^2 + U_{pd}^g k_{ppll}s + U_{pd}^g k_{ipll}}.$$
(17)

Because in the synchronous reference frame  $U_{pq}^g = 0$  and assuming the VSC is operating in the unity power factor, (15) can be further simplified to (18) as follows:

$$\begin{bmatrix} \widetilde{u}_{pd}^c\\ \widetilde{u}_{pq}^c \end{bmatrix} \approx \begin{bmatrix} 0\\ -U_{pd}^g \Delta \theta + I_d \Delta \theta \end{bmatrix}.$$
 (18)

From (18), we can conclude that the dynamics introduced by the operation of the PLL during perturbations add  $-U^{g}_{pd} \Delta \theta$  and  $+ I_{d} \Delta \theta$  terms in the *q*-axis control loop, as depicted in Figure 5. On the other hand, due to the coupling between two axes, any variations in  $\Delta \theta$  leads to the changes in  $i_d$  (this correlation is elaborated in Section 4).

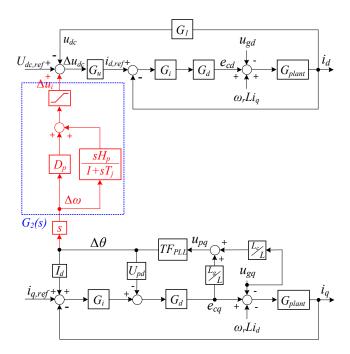


Figure 5. The *d*-*q* control loops augmented with the frequency support controller.

#### 3.3. Fast Grid Frequency Support

The imbalance between generation and demand following a contingency would cause the frequency to fall/rise at a rate determined by the total grid inertia. The less inertia, the faster and higher frequency deviations. Various methodologies for providing synthetic inertia have been reported, which further information can be found in [27]. A promising technique recommended in [21], in which the frequency support is conducted by means of regulating the DC-link capacitor voltage. This method is based on a basic proportional controller that bounds its responsiveness [22]. Herein, this problem is fixed using a supplementary controller composed of a frequency-time derivative (df/dt) term (aimed at mitigation of the RoCoF) with a scheduled active power injection and delivery time  $T_j$ . The time delay  $T_j$  in discharging the DC-link capacitor is analogous to gradually releasing the kinetic energy of rotational SGs. The controller is added to the outer voltage loop during the initial support period. The proposed technique is described mathematically as:

$$u_i(s) = D_p(\omega - \omega_r) + \frac{H_p s \omega}{1 + sT_j}$$
(19)

where  $H_p$  and  $D_p$  emulate the inertia and damping characteristics of the real SGs, respectively. A disturbance in the power system results in a small deviation in the variables around their steady state values. Thus, the corresponding small-signal model of the synthetic inertia loop can be derived by canceling the steady state values of (19) as:

$$\Delta u_i(s) = D_p \Delta \omega + \frac{H_p s \Delta \omega}{1 + sT_j}$$

$$G_2(s) = \frac{\Delta u_i(s)}{\Delta \omega(s)} = D_p + \frac{H_p s}{1 + sT_j}.$$
(20)

It is noteworthy that  $\Delta u_i = 0$  during steady state operation of the system. Indeed, the frequency support level is limited by the maximum acceptable DC-link voltage deviations in the foregoing method. This is because  $u_{dc}$  must be kept in an acceptable range for linear modulation of the converter and limiting current flow through components of the conversion system. Thereby, the *d*-*q* control loops augmented with the frequency support controller are formed as Figure 5.

#### 4. Dynamics Assessment of the Proposed Control Technique

To validate the effectiveness of the proposed control system, it is necessary to analyze the stability of the control loops. According to Figure 5, the closed-loop transfer functions of inner current loops corresponding to the *d*-*q* axes can be obtained as (21) and (22).

$$TF_{id}(s) = \frac{i_d(s)}{i_{d,ref}(s)} = \frac{G_i(s) \ G_d(s) \ G_{plant}(s)}{1 + G_i(s) \ G_d(s) \ G_{plant}(s)}$$
(21)

$$TF_{iq}(s) = \frac{i_q(s)}{i_{q,ref}(s)} = \frac{L \ G_i(s) \ G_d(s) \ G_{plant}(s)}{L + L_g \ U_{pd} \ G_d(s) \ TF_{PLL}(s) - L_g \ I_d \ G_i(s) \ G_d(s) \ TF_{PLL}(s) + L \ G_i(s) \ G_d(s) \ G_{plant}(s)}.$$
 (22)

Because of the coupling between the *d*-*q* axes, there is a correlation between  $\Delta\theta$  and  $i_d$ , which produces an effect on the outer voltage controller by the synthetic inertia loop. This relationship is visible in Figure 5. Hence, the transfer function between  $\Delta\theta$  and  $i_d$  is derived as (23).

$$TF_{\Delta\theta/id}(s) = \frac{\Delta\theta(s)}{i_d(s)} = \frac{\omega_r L_g L G_i(s) G_d(s) G_{plant}(s) TF_{PLL}(s)}{L + L_g U_{pd} G_d(s) TF_{PLL}(s) - L_g I_d G_i(s) G_d(s) TF_{PLL}(s) + L G_i(s) G_d(s) G_d(s) G_{plant}(s)}.$$
 (23)

In addition, the frequency support can be provided by incorporating the angular frequency deviations ( $\Delta \omega$ ) into the DC-link voltage regulator. Mathematically,  $\Delta \omega$  is equivalent to the time-derivative of  $\Delta \theta$ , as illustrated in Figure 4b. Thereby, the transfer function between  $\Delta u_i$  and  $i_d$  is derived as:

$$TF_{\Delta ui/id}(s) = \frac{\Delta u_i(s)}{\Delta \omega(s)} \times \frac{\Delta \omega(s)}{\Delta \theta(s)} \times \frac{\Delta \theta(s)}{i_d(s)} = s \ G_2(s) \ TF_{\Delta \theta/id}(s) \ .$$
(24)

According to Figure 5, the transfer function between  $\Delta u_i$  and  $i_d$  can be considered in parallel with  $G_1$ . This is because the two corresponding paths start with  $i_d$  and reach the identical summer. It means that the dynamics caused by the PLL in the *q*-axis are reflected in the *d*-axis by means of the virtual inertia loop. Then, the function that appeared in the feedback path is formed as follows:

$$G_2(s) = G_1(s) - TF_{\Delta ui/id}(s).$$
<sup>(25)</sup>

Finally, the open-loop and closed-loop transfer functions of the proposed control system are derived as follows, respectively:

$$(TF_{udc/\Delta udc})_{ol} = \frac{u_{dc}(s)}{\Delta u_{dc}(s)} = G_u(s) \ TF_{id}(s) \ G_2(s)$$
(26)

$$(TF_{udc/\Delta udc})_{cl} = \frac{u_{dc}(s)}{U_{dc}(s)} = \frac{G_u(s) \ TF_{id}(s) \ G_1(s)}{1 + G_u(s) \ TF_{id}(s) \ G_1(s) \left(1 - \frac{TF_{\Delta ui/id}(s)}{G_1(s)}\right)}.$$
(27)

The stability of the system is investigated by bode diagram and zero-pole map analyses. Figure 6 depicts the bode-plot of the frequency response of the open-loop model, with the parameters listed in Table 2. As can be seen from this figure, the gain and phase margins are 45 dB and 27.2 deg, respectively, which validate the stable operation of the model described by (26). Moreover, the zero-pole diagram of the closed-loop transfer function (27) is demonstrated in Figure 7. The zeros and poles are represented by circles and crosses, respectively. It is clear that all the poles are located in the left-hand side of the complex plane, which signifies the stability of the whole system. Therefore, the proposed controller is mathematically proved to operate stably.

To demonstrate the advantage of the proposed controller compared to [21] in terms of stability margin, the same bode analysis has been carried out for [21], and the result is illustrated in Figure 8. As observed from this figure, the gain margin and phase margin of the VSC system controlled by the method in [21] drop to 4.33 dB and 20.4 deg, respectively. This deterioration in the stability metrics arises from only applying the df/dt term and its corresponding gain in [21], in which this problem is

solved in the proposed control scheme using transfer function  $G_2(s)$  (Figure 5). Thus, the proposed controller has better performance in comparison to [21] regarding the system stability margin.

System Parameters	Values	<b>Controllers Parameters</b>	Values
SG nominal power	100 kW	H, D	3, 1
VSC nominal power	15 kW	$H_p$ , $D_p$	50, 100
Lg	0.003 H	k <sub>cp</sub>	0.1
$L_c$	0.002 H	k <sub>ci</sub>	10
r	0.001 Ω	$k_{up}$	0.006
$U_{pd}$	326.59 V	k <sub>ui</sub>	0.001
$U_{dc,ref}$	750 V	k <sub>ppll</sub>	180
C	0.1 F	$k_{ipll}$	3200
ω <sub>r</sub>	$100\pi$ rad/s	$T_j$	0.2

Table 2. The specifications of the implemented system.

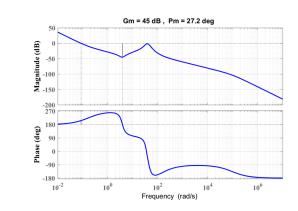


Figure 6. Bode diagram of the proposed control system.

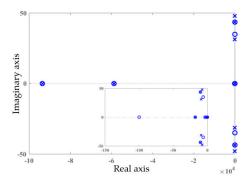


Figure 7. Zero-pole diagram of the proposed control system.

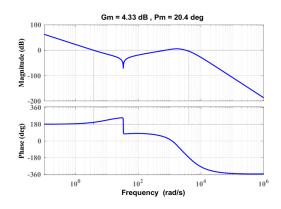


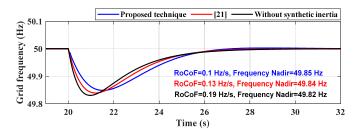
Figure 8. Bode diagram of the method in [21].

#### 5. Simulation Results and Discussions

Simulations have been conducted in MATLAB to demonstrate the validity of the proposed technique for control of the grid-tied VSCs. The main parameters of the implemented system are tabulated in Table 2. A 100 kW synchronous generator models the power grid with the internal impedance of  $0.001 + j0.94 \Omega$ . The grid-tied VSC operates under the following conditions. The rated power is 15 kW. The maximum acceptable DC-link voltage deviation is ±60 V, and the modulation index is considered  $m_a = 0.86$ . Two scenarios are considered for simulations, step-up and step-down changes in the demand. Then, in each scenario, the proposed controller is compared with the method in [21] and the conventional VSC system in terms of frequency stability metrics (i.e., frequency nadir and RoCoF).

## 5.1. Scenario 1

In this scenario, a 5% step-up change in the demand occurs at t = 20 s. Figure 9 depicts frequency variations when the grid-interactive converter is controlled by the (1) proposed controller, (2) method in [21], and (3) conventional control system without virtual inertia functionality. As observed from this figure, without providing synthetic inertia, the frequency nadir and RoCoF peaks at 49.82 Hz and 0.19 Hz/s, respectively, and the grid frequency is stabilized after 6 s. The VSC system augmented with the proposed virtual inertia loop can improve the RoCoF level by 47.37% and 23.1% compared to the conventional controller and [21], respectively. Moreover, the proposed technique yields lower maximum frequency oscillations (i.e., 49.85 Hz) in comparison to the other ones. It should be mentioned that further providing inertia support by the method in [21] leads to the instability of the whole system, which is in agreement with the stability analysis presented in Figure 8. Thus, better primary frequency regulation during transient time is achieved with the proposed supplementary controller. The DC-link capacitor voltage oscillation is illustrated in Figure 10. The voltage drops to 690 V right after the step change in the load and returns to the nominal value (750 V). It means that the discharged energy by the DC-link capacitor supports the frequency, which is analogous to the released kinetic energy of SGs during perturbation.



**Figure 9.** Frequency nadir and rate of change of frequency (RoCoF) metrics under a 5% step-up change in the demand.

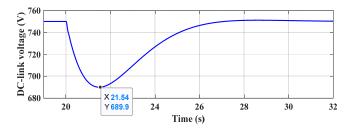


Figure 10. DC-link voltage oscillation under a step-up change in the demand.

As stated earlier, the DC-link voltage deviations must be limited, which means to a finite frequency support level. When the DC-link voltage is reduced with the aim of frequency support, the modulation index ( $m_a$ ) must increase in order to (1) increase the difference angle between converter voltage and

PCC voltage ( $\Delta$ ) and (2) prevent reactive power exchange with the grid (i.e., the equation  $e_d = m_a \times u_{dc} \times cos\Delta = u_{pd}$  holds true). This can be observed in Figure 11. The modulation index peaks at around 0.93 in the time interval at which the capacitor is discharged. Then, it rebounds to the value of pre-perturbation. The supportive power supplied by the VSC is shown in Figure 12. It is clear that the power reaches approximately 15 kW and then drops to the steady state value when the frequency returns to 50 Hz.

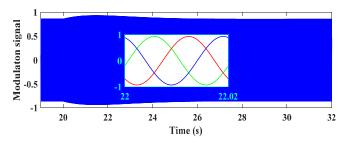


Figure 11. Modulation signal waveform in the first scenario.

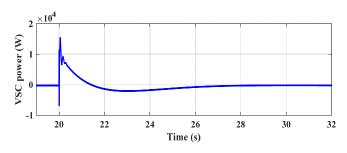


Figure 12. Active power exchange with the grid in the first scenario.

### 5.2. Scenario 2

In this case study, the demand drops by 5% at t = 20 s. Figure 13 illustrates frequency deviations of the system with (1) the proposed control scheme, (2) the method in [21], and without providing synthetic inertia. It is clear that the frequency nadir peaks at 50.17 Hz when the synthetic inertia loop is canceled out. The RoCoF level is also 0.19 Hz/s. Taking the proposed supplementary controller into account can effectively improve the grid frequency stability and bring the frequency back to the reference (50 Hz). In this case, the rate of change of frequency is enhanced by 47.37% and 23.1% compared to the conventional controller and [21], respectively. Thus, the obtained results validate the effectiveness of the proposed control technique in enhancing frequency stability following a disturbance. The DC-link capacitor voltage deviation of this scenario is also illustrated in Figure 14. Clearly, the voltage rises to 810 V right after the loss of demand and reaches the steady state value after around 6 s.

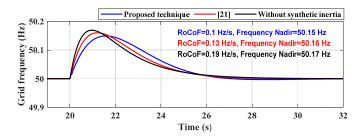
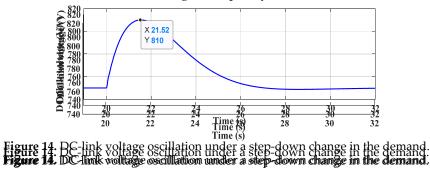


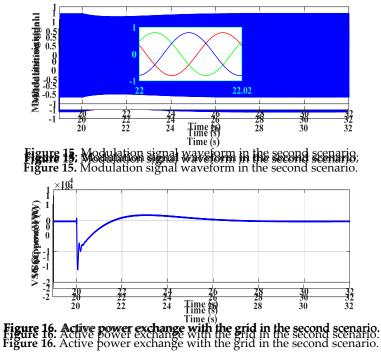
Figure 13. Frequency nadir and RoCoF metrics under a 5% step-down change in the demand.

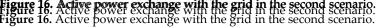
The same analysis can be carried out for the decrease in modulation index when the DC-link voltage starts to rise. This is because the difference angle  $\Delta$  must increase while the reactive power

contingency (again, the equation  $e_d = m_a \times u_{dc} \times cos\Delta = u_{pd}$  holds true). VIEW VIEW VIEW VIEW Set 12 VIEW Set 1 exchange i Sustainability Sustainability Sustainability remains z y 2020, 12, x y 2020, 12, x y 2020, 12, x a) in this scenario is observed in Figure 15. The power absorbed by the VSC uency support is also brought in Figure 16. The power reaches approximately uency support is also brought in Figure 16. The power reaches approximately of demand and becomes zero when the grid frequency returns to 50 Hz. -15 kW at









# 6. Conclusions

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# Nomenclature

Abbreviations:	
dq <sup>g</sup> , dq <sup>c</sup>	Synchronous reference frames corresponding to the grid and the controller
ESS	Energy storage system
PLL	Phase-locked loop
PCC	Point of common coupling
RESs	Renewable energy sources
RoCoF	Rate of change of frequency
SGs	Synchronous generators
VSC	Voltage source converter
VSG	Virtual synchronous generator
Variables:	, ,
i	Current flowing into the grid
ma	Modulation Index
$P_m$	Input mechanical power of SG
$P_e$	Electromagnetic power of SG
$u_{dc}, e_c, u_p, u_g$	DC-link capacitor voltage, converter voltage, PCC voltage, grid voltage
$\underline{u}_{p}^{g}, \underline{u}_{p}^{c}$	PCC voltage vectors in the grid frame and the controller frame
ω	Angular velocity/frequency of the rotor/grid
Δ	The difference angle between the converter voltage and the PCC voltage
Parameters:	
С	DC-link capacitance
$D, D_p$	Damping coefficient, virtual damping coefficient
F <sub>HP</sub>	Turbine coefficient
$f_s$	Sampling frequency
H, H <sub>p</sub>	Inertia constant, virtual inertia constant
I <sub>d</sub>	Rated value of the current <i>i</i> in the <i>d</i> -axis
J	Moment of inertia of the turbine and the generator
k <sub>cp</sub> , k <sub>ci</sub>	Proportional and integral gains of the current controller
k <sub>up</sub> , k <sub>ui</sub>	Proportional and integral gains of the voltage controller
k <sub>ppll</sub> , k <sub>ipll</sub>	Proportional and integral gains of the PLL
$L_c, L_g$	Inductance of the filter and the grid
$R_c, R_g$	Resistance of the filter and the grid
R	Droop coefficient
$T_G$	Governor time constant
$T_{CH}$	Inlet volume time constant
$T_{RH}$	Re-heater time constant
U <sub>dc,ref</sub>	DC-link voltage reference
$U_{pd}$	Nominal PCC voltage in the <i>d</i> -axis
$\omega_r$	Reference angular velocity/frequency of the rotor/grid

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